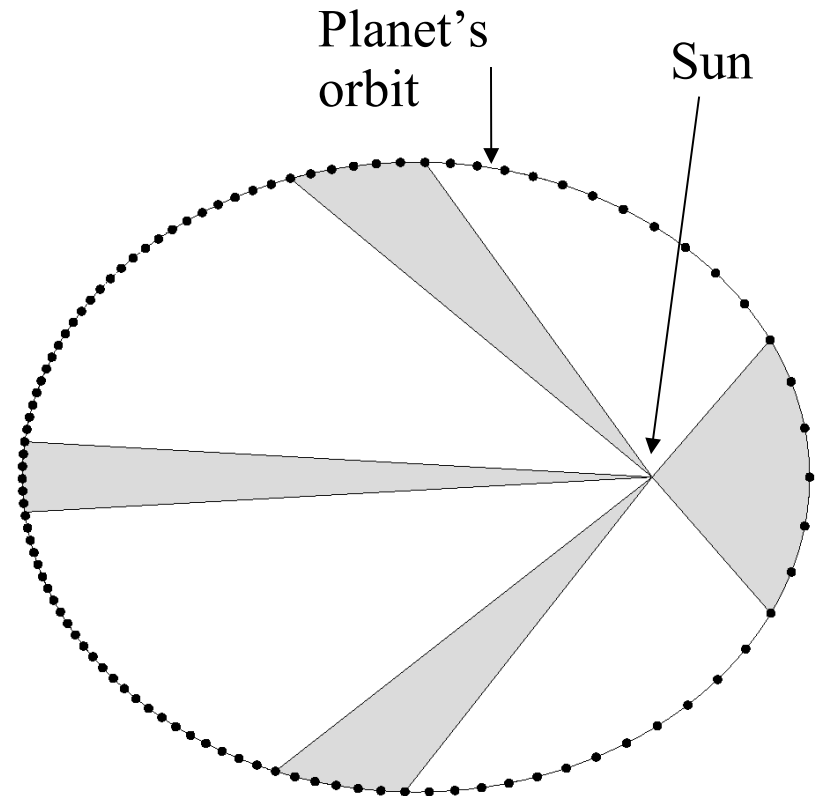


Kepler's Laws of Motion

- 1609 in *Astronomia Nova* (The New Astronomy)
- First Law – A planet orbits the Sun in an ellipse, with the Sun at one focus of the ellipse.
- Second Law – A line connecting a planet to the Sun sweeps out equal areas in equal time intervals
 - Several areas associated with the time interval of “six” are shown
 - They all have equal areas



Kepler's Third Law of Motion

From *Harmonia Mundi* (1619) (Harmony of the Worlds)

$$P^2 = a^3$$

P = orbital period

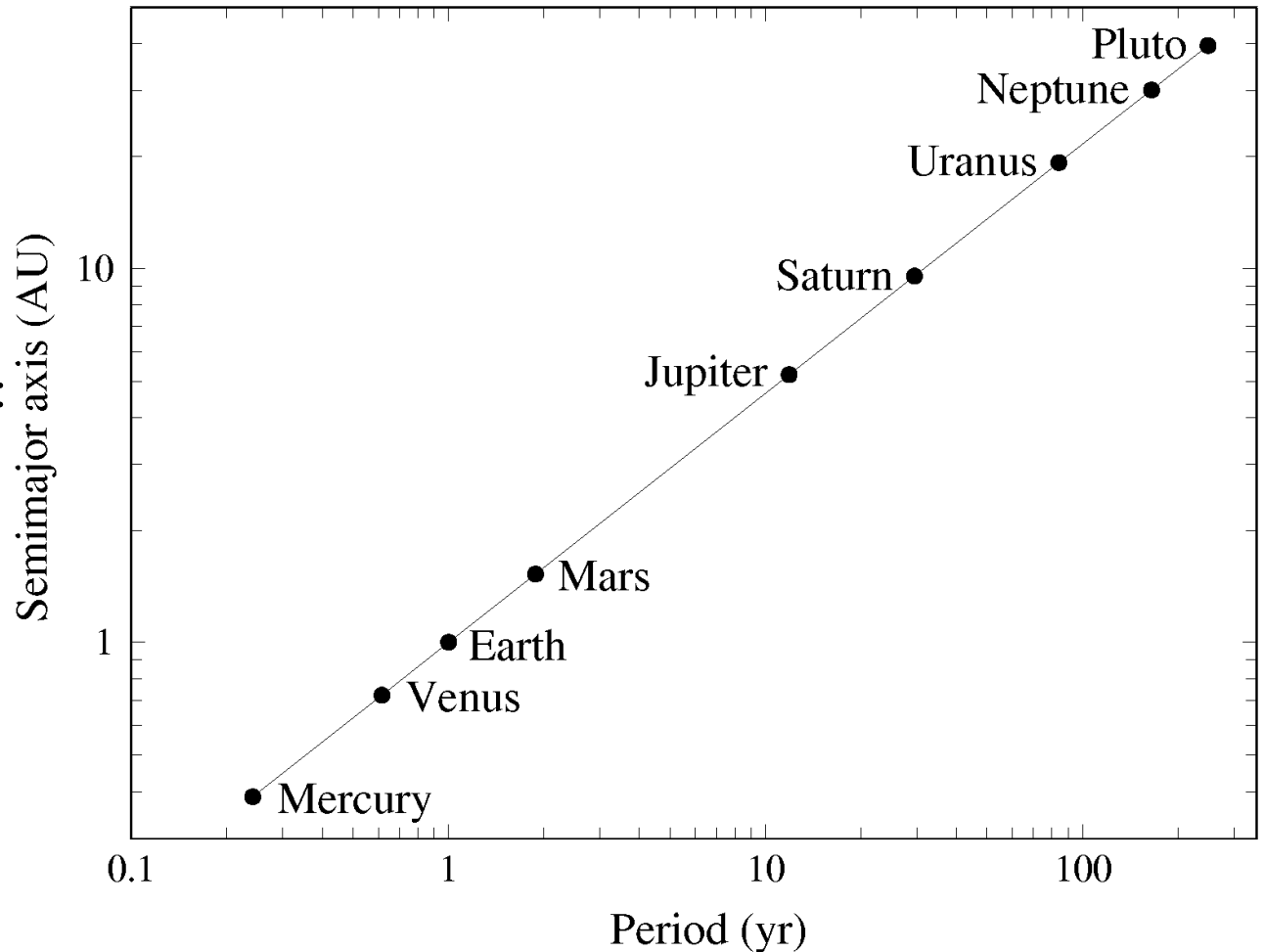
a = semimajor axis

“Power law” slope is 2/3:

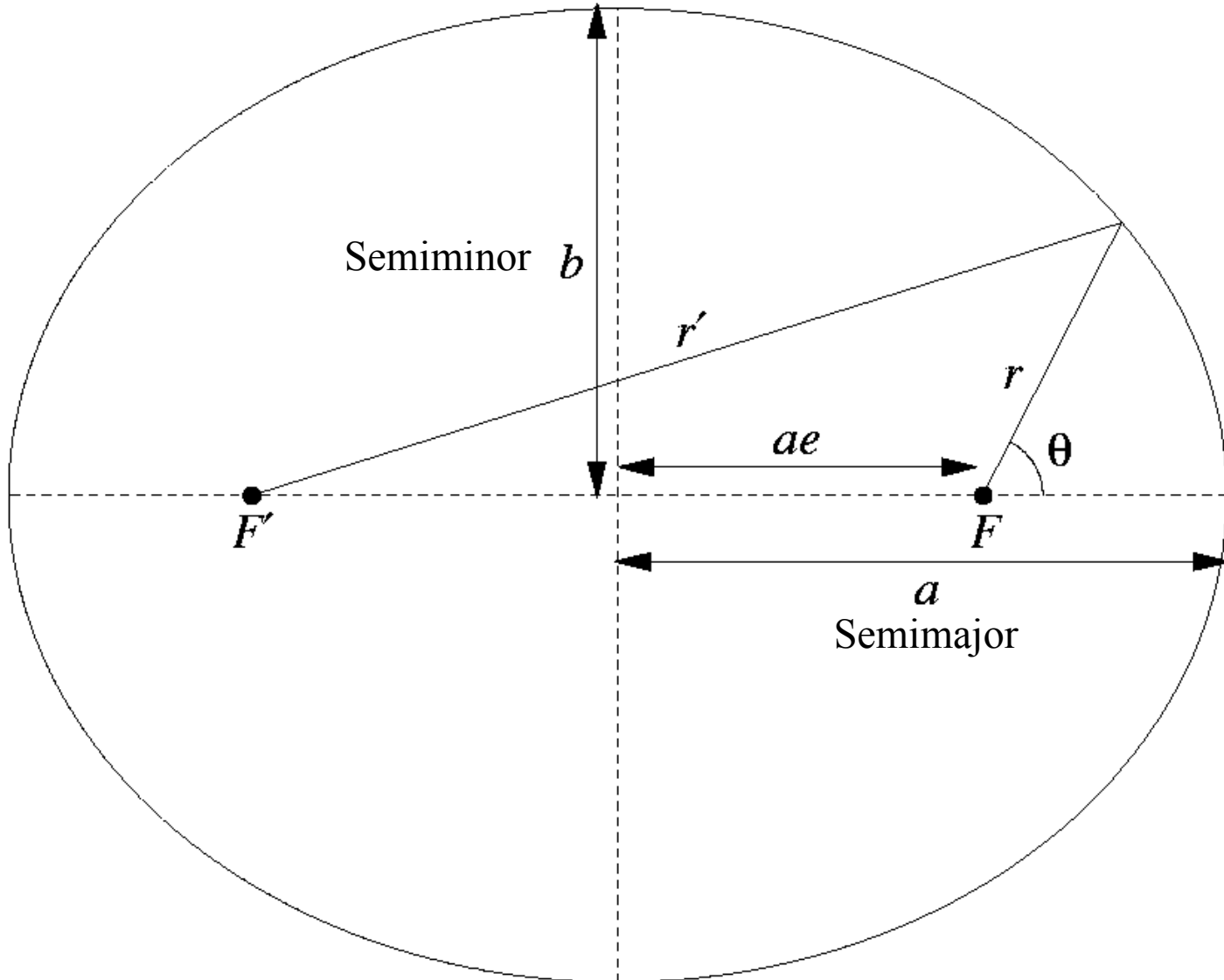
$$\log(P^2) = \log(a^3)$$

$$2\log(P) = 3\log(a)$$

$$\log(a) = \frac{2}{3}\log(P)$$

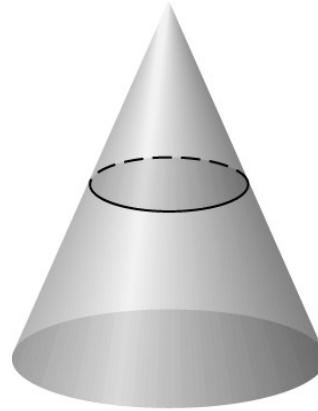


Ellipses

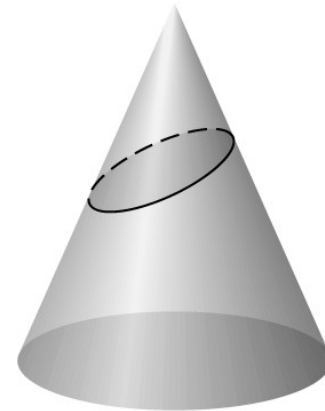


Conic Sections

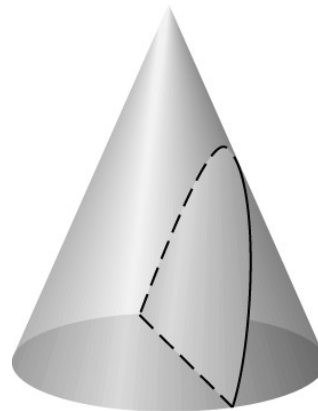
- Intersection of a plane with a cone
- Parabola – plane is parallel to a side
- Hyperbola – plane is parallel to central axis
- All are possible orbits (elliptical orbits most common)



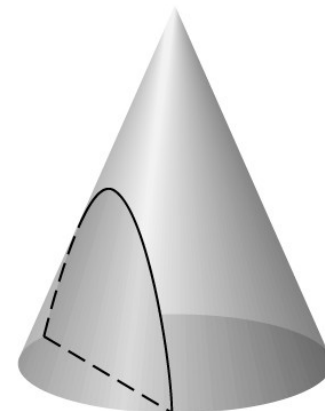
Circle



Ellipse



Parabola



Hyperbola

Conic Sections

- All are possible in celestial mechanics.
- “p” is closest approach for parabolic orbit

$$r = a \quad e = 0 \quad \text{Circle}$$

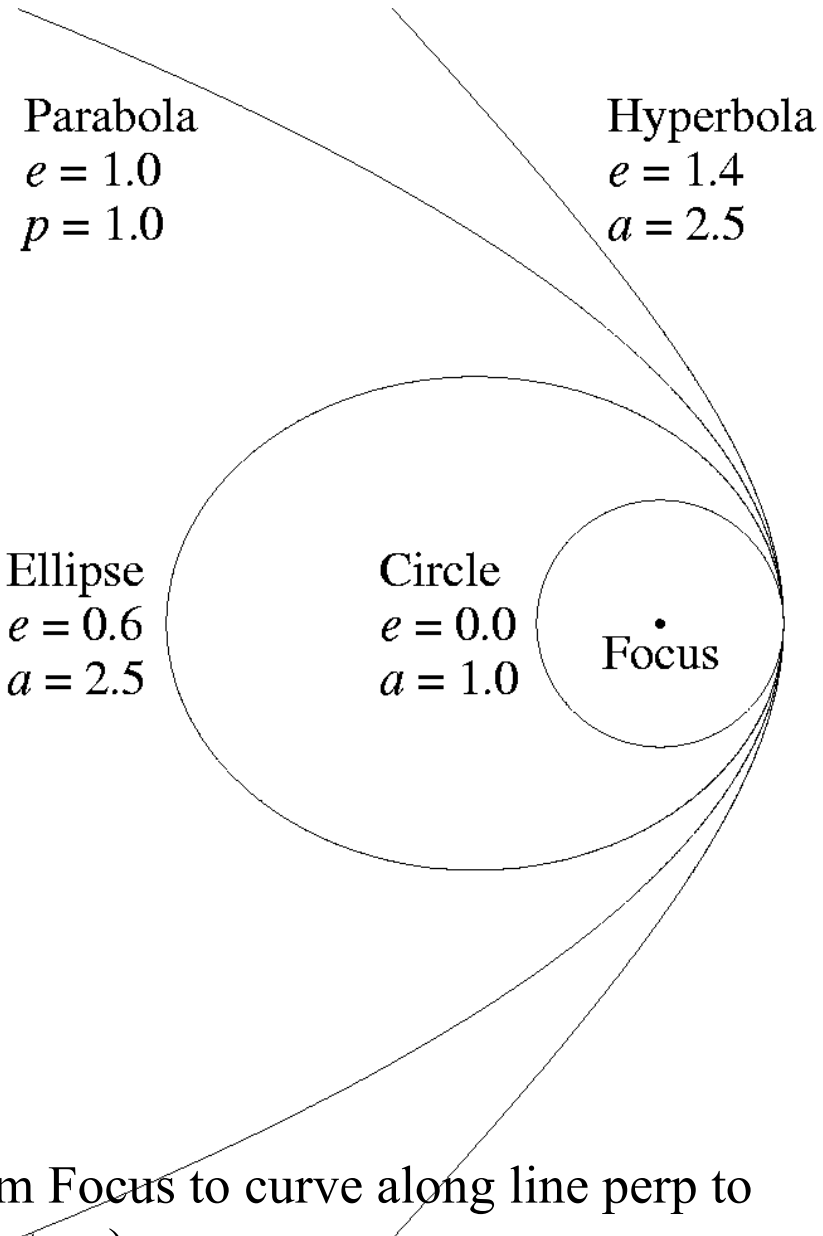
$$r = \frac{a(1-e^2)}{1+e \cos \theta} \quad 0 \leq e < 1 \quad \text{ellipse}$$

$$r = \frac{2p}{1+\cos \theta} \quad e = 1 \quad \text{parabola}$$

$$r = \frac{a(e^2-1)}{1+e \cos \theta} \quad e > 1 \quad \text{hyperbola}$$

$$r = \frac{L}{1+e \cos \theta}$$

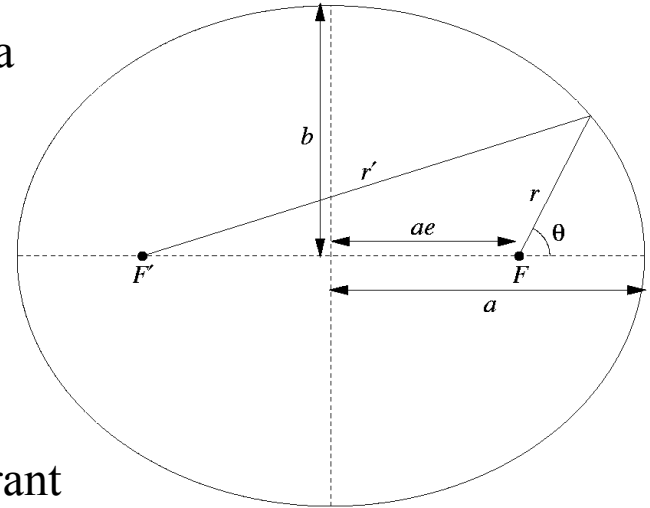
General, where L = dist from Focus to curve along line perp to major axis (semi-latus rectum)



Ellipse Drawing

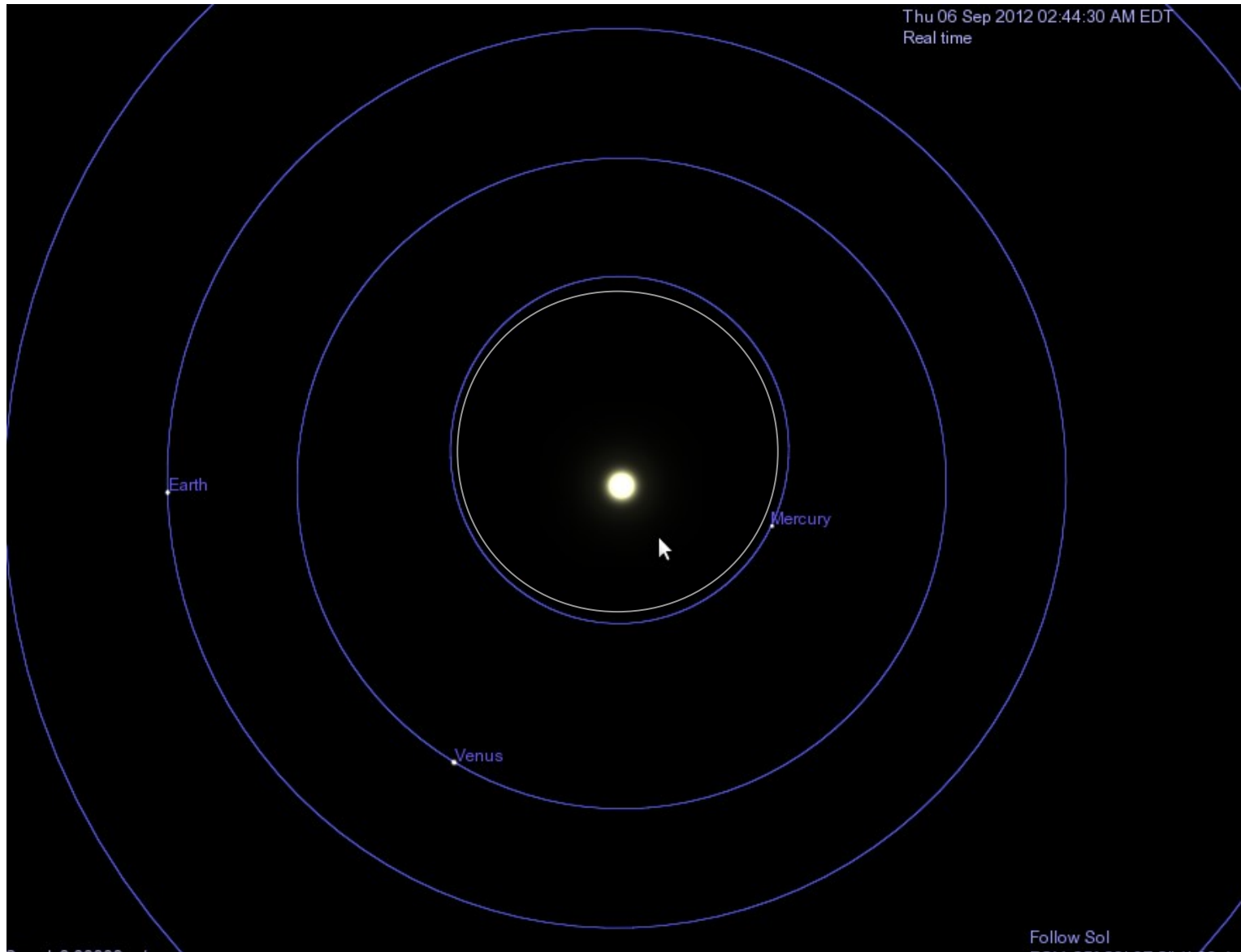
After drawing your ellipse on graph paper by keeping a pencil snug against a string looped loosely around two tacks, do the following:

- 1) Mark center "O".
- 2) Mark F and F' (foci).
- 3) Measure and label a and b (in mm).
- 4) Measure and label ae.
- 5) Draw point (labelled "P") on ellipse in the 1st quadrant position. Draw, label, and measure r and r'. Measure θ .
- 6) Confirm $r + r' = 2a$
- 7) Calculate eccentricity using $e = ae/a$
- 8) Calculate eccentricity using $e = \sqrt{1 - \left(\frac{b}{a}\right)^2}$
- 9) Confirm that $r = a(1 - e^2)/(1 + e\cos\theta)$
- 10) Measure x and y for P, where $(x,y) = (0,0)$ at center (not focus)
- 11) Confirm the Cartesian coordinate equation for the ellipse using point P: $\frac{x^2}{a^2} + \frac{y^2}{b^2} = 1$



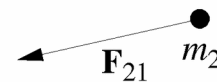
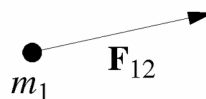
Ellipses – actual orbits

(September 2012)



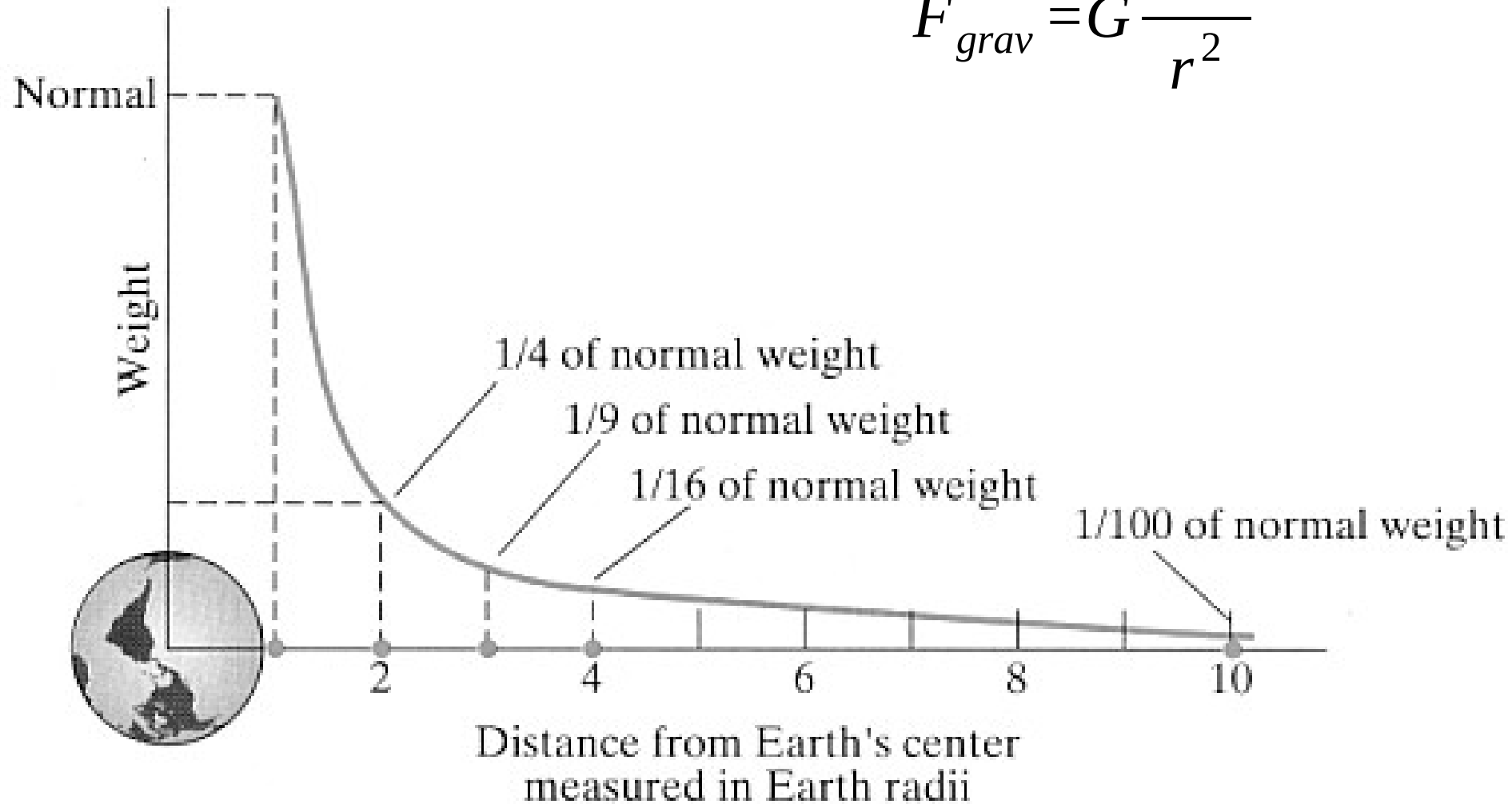
Newton's Laws of Motion

- 1st Law – Law of inertia
 - An object at rest remains at rest and an object in uniform motion remains in uniform motion unless acted upon by an unbalanced force.
 - An *inertial reference frame* is needed for 1st law to be valid
 - A non-inertial reference frame is accelerating (e.g. In car going around a curve you feel a fictitious force)
- 2nd Law – $\mathbf{a} = \mathbf{F}_{\text{net}}/m$ or $\mathbf{F}_{\text{net}} = m\mathbf{a}$
 - The net force (sum of all forces) acting on an object is proportional to the object's mass and its resultant acceleration.
 - Inertial mass, m , does not appear to be different from gravitational mass
- 3rd Law
 - For every action there is an equal but opposite reaction



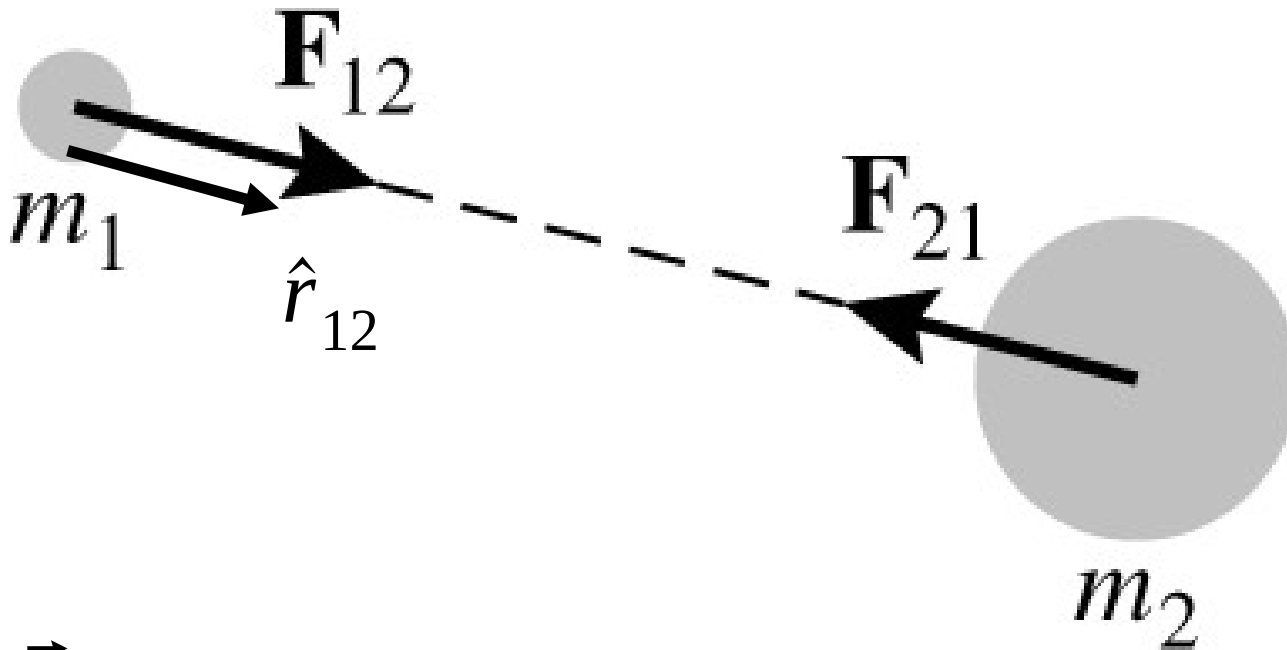
Universal Law of Gravitation

$$F_{grav} = G \frac{Mm}{r^2}$$



Universal Law of Gravitation

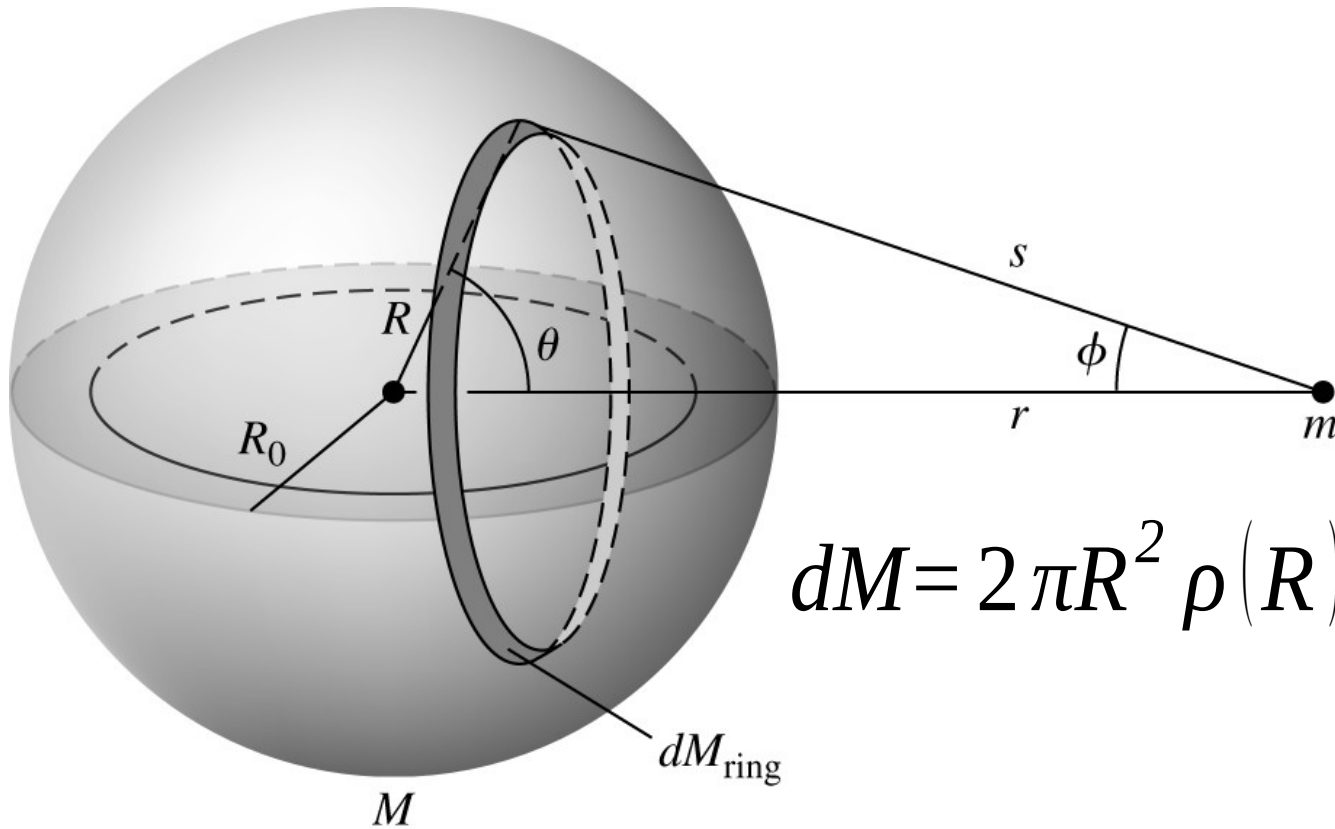
$$\vec{F}_{12} = G \frac{Mm}{r^2} \hat{r}_{12}$$



\vec{F}_{12} is force on 1 by 2.
Unfortunately, this is opposite the convention used in PHYS 2321 (Coulomb's Law)

Shell theorems for gravity:

-) The Force on m due to a uniform shell of mass is the same as the force due to a point mass at the center of the shell with the same total mass as the shell.
-) The force of gravity inside of a uniform shell is zero.



$$dM = 2\pi R^2 \rho(R) \sin\theta dR d\theta$$

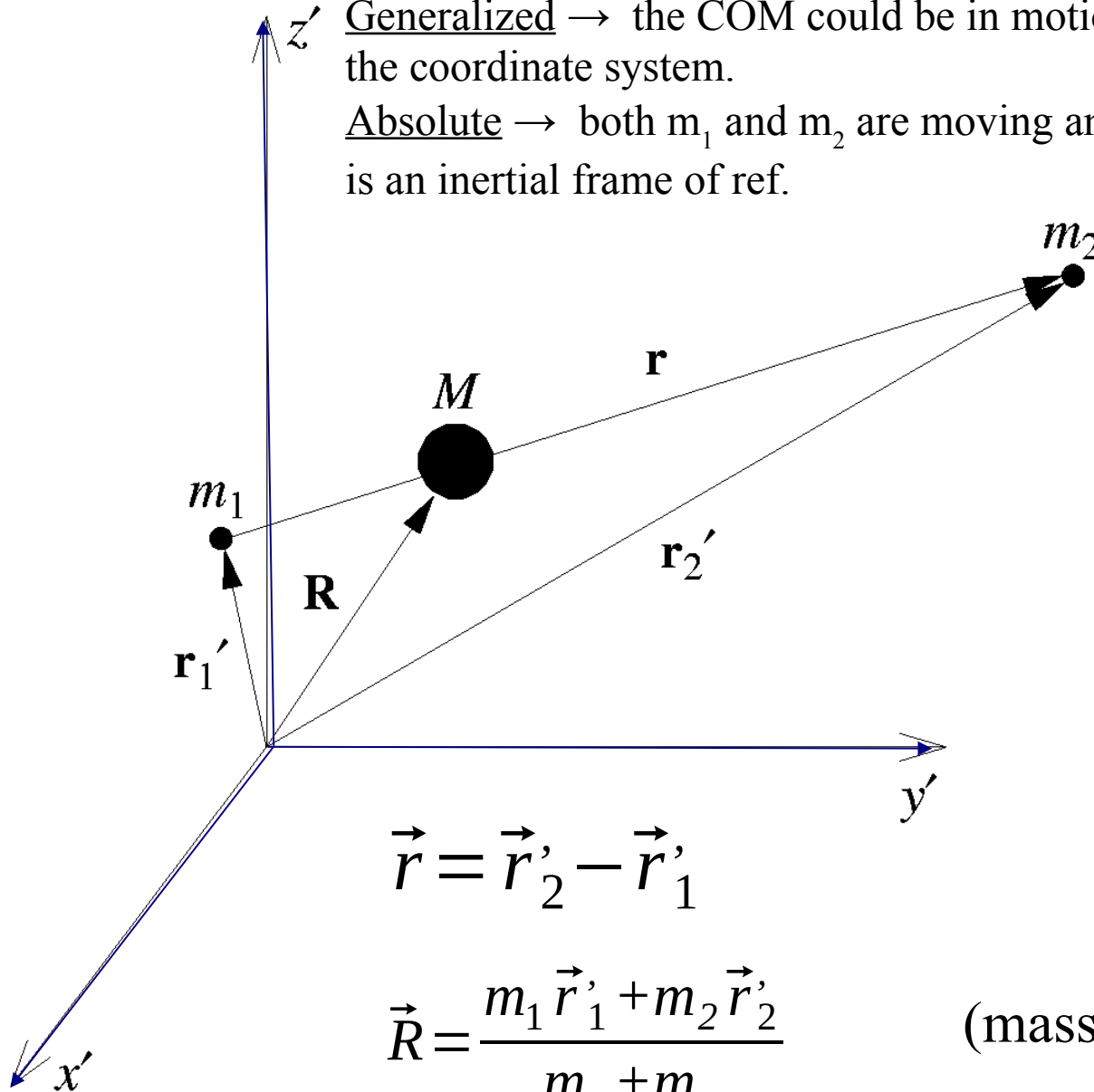
(See Ch. 2 derivation of $F_{\text{shell}} = GM_{\text{shell}}m/r^2$.)

Binary Orbits

Generalized, absolute coordinates.

Generalized → the COM could be in motion relative to the coordinate system.

Absolute → both m_1 and m_2 are moving and the coord sys is an inertial frame of ref.



$$\vec{r} = \vec{r}'_2 - \vec{r}'_1$$

$$\vec{R} = \frac{m_1 \vec{r}'_1 + m_2 \vec{r}'_2}{m_1 + m_2}$$

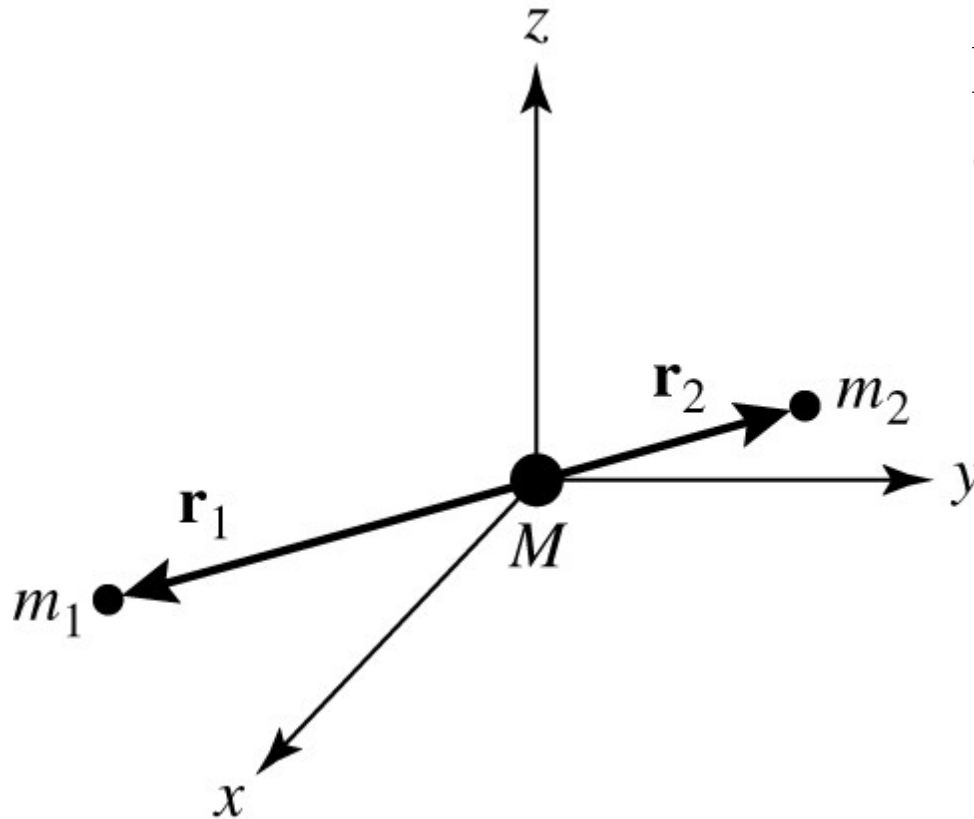
(mass-wtd average)

Binary Orbits

Absolute coordinates.

Absolute → both m_1 and m_2 are moving and the coord sys is an inertial frame of ref.

The COM is placed at the origin. It is labeled with the total mass $\mathbf{M} = \mathbf{m}_1 + \mathbf{m}_2$.



M is closer to the bigger mass (here m_2).

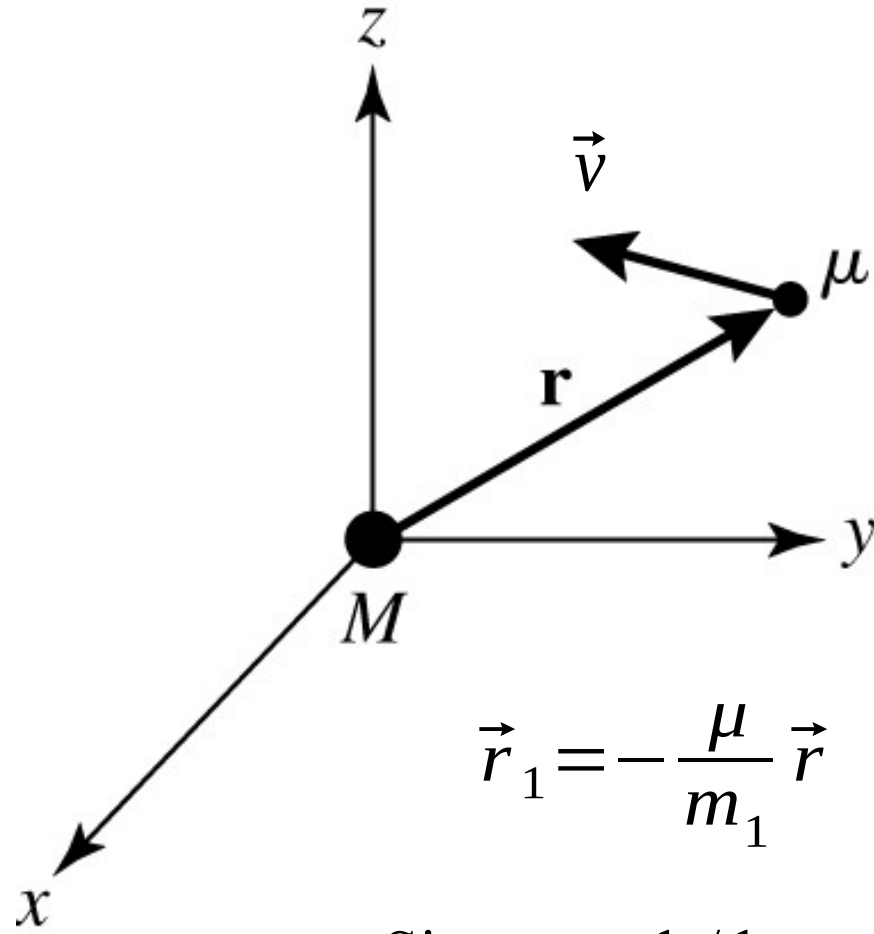
$$m_1 r_1 = m_2 r_2 \text{ SO}$$

$$r_1 / r_2 = m_2 / m_1$$

Binary Orbits

Relative coordinates.

Relative → shows orbit of moving, *reduced mass* μ around a stationary *total mass* M .



$$\vec{r} = \vec{r}'_2 - \vec{r}'_1$$

$$|\vec{r}| = r_1 + r_2$$

$$\mu = \frac{m_1 m_2}{m_1 + m_2} = \frac{m_1 m_2}{M}$$

$$\vec{r}'_1 = -\frac{\mu}{m_1} \vec{r}$$

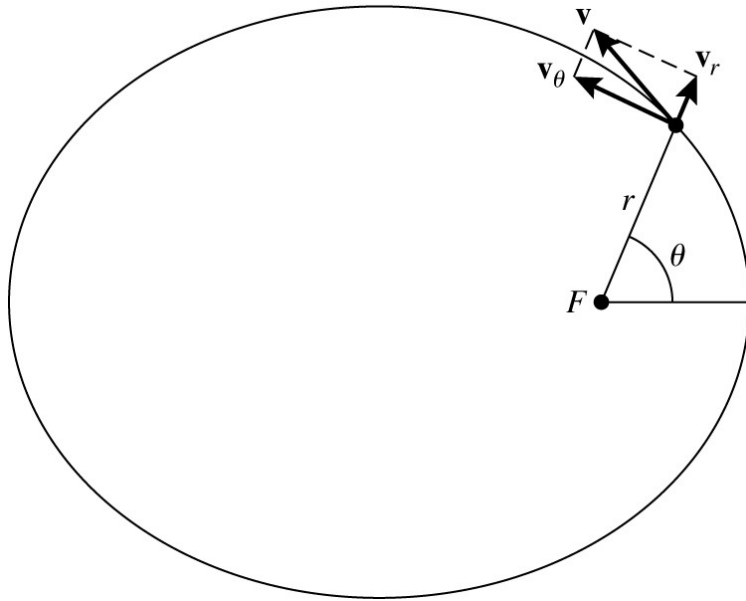
$$\vec{r}'_2 = \frac{\mu}{m_2} \vec{r}$$

Since $\mathbf{v} = d\mathbf{r}/dt$, $\mathbf{v}_1 = -\mu/m_1 \mathbf{v}$, and $\mathbf{v}_2 = \mu/m_2 \mathbf{v}$

Binary Orbits

Absolute coordinates and velocity.

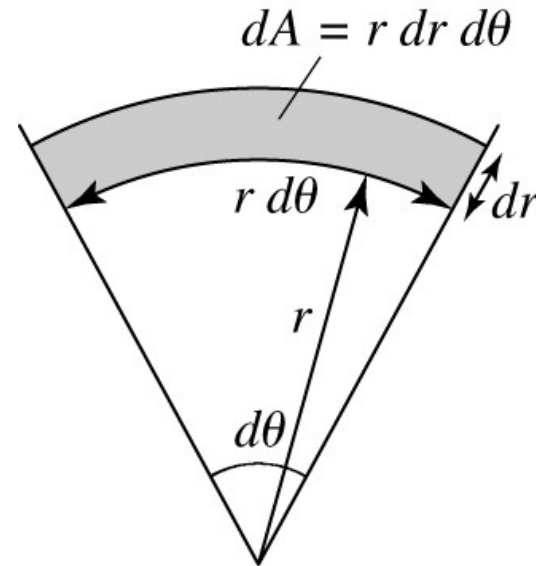
Velocity vector is only purely tangential at perihelion and aphelion.



$$v_{\theta}^2 + v_r^2 = v^2$$

$$v_{\theta} = r \frac{d\theta}{dt}$$

$$v_r = \frac{dr}{dt}$$

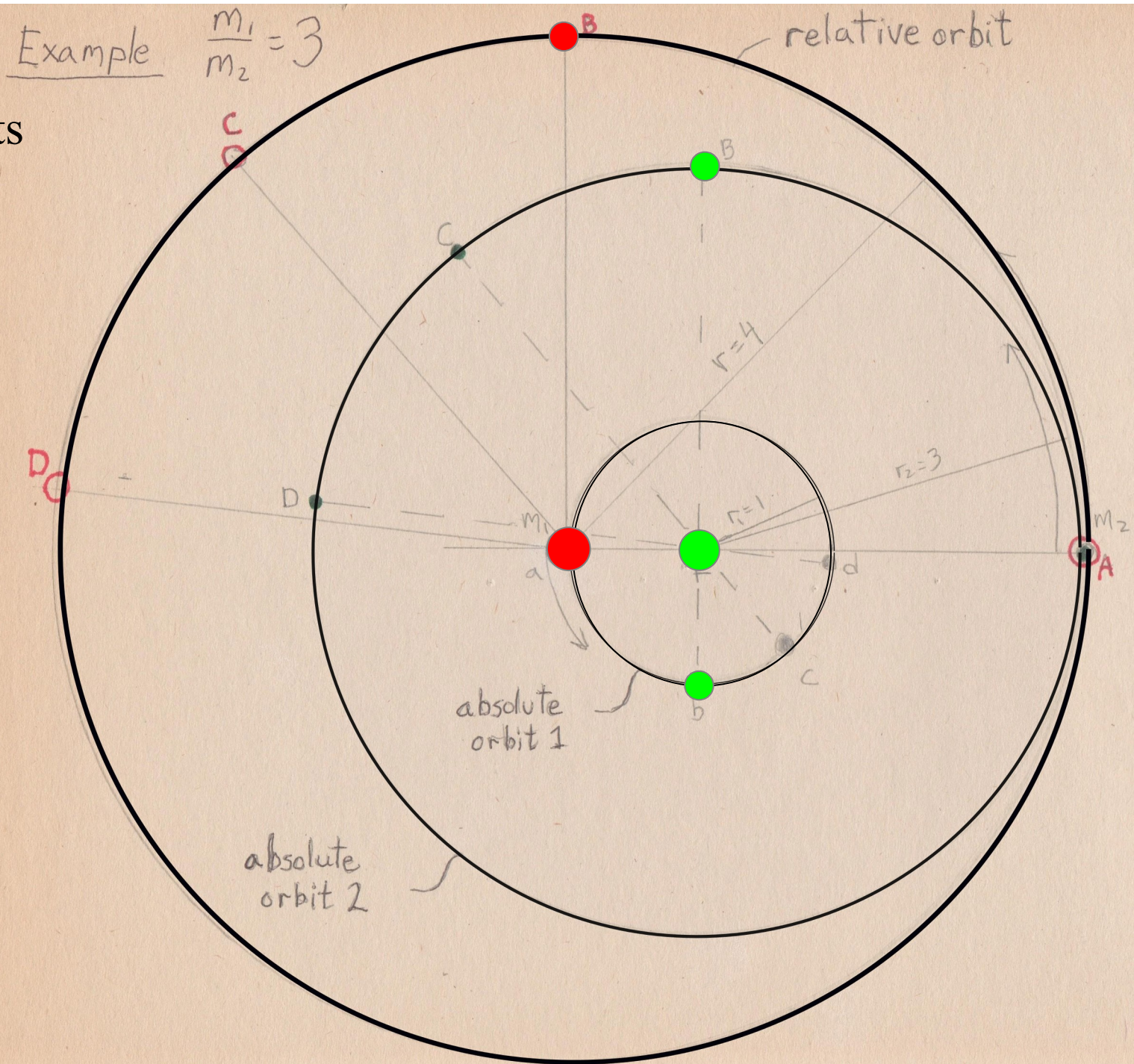


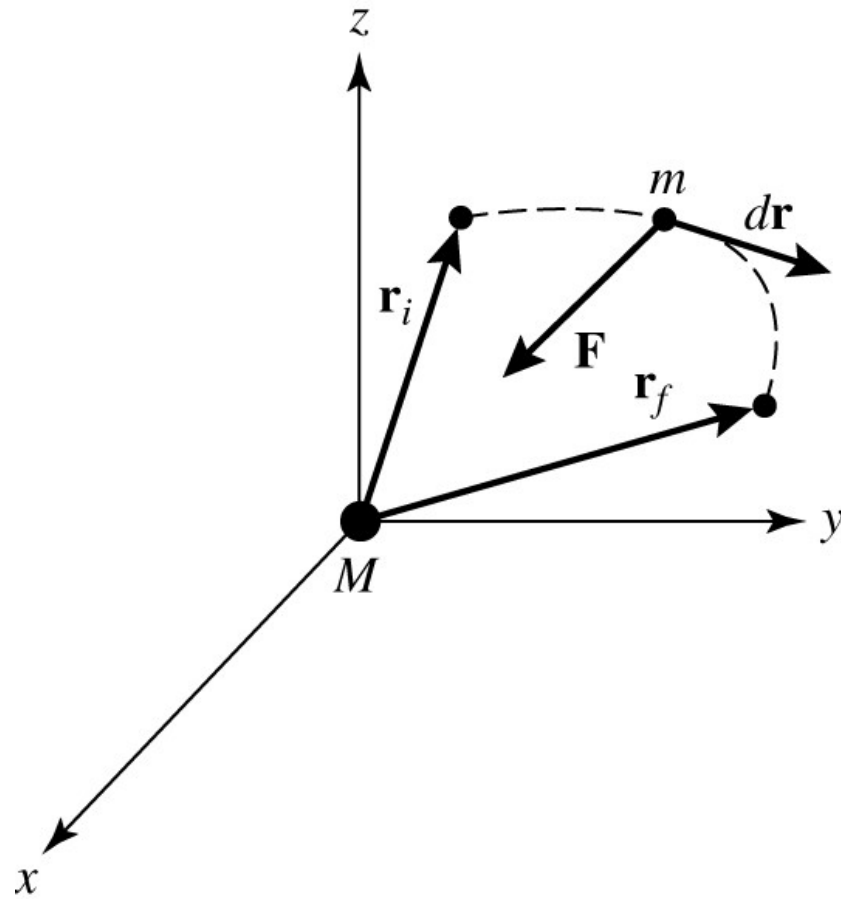
$$dA_{\text{sector}} = \int_0^R r dr d\theta$$

$$dA_{\text{sector}} = \left[\frac{R^2}{2} - 0 \right] d\theta$$

$$dA_{\text{sector}} = \frac{R^2}{2} d\theta$$

Binary Orbits





Work by gravity depends on direction of net force vector relative to the direction of motion.